ABSTRACT OF THE INVENTION

A tube-type vortex reducer for the conduction of cooling air in a compressor (1) of a gas turbine, with radial secondary air tubes (2) being arranged in a disk interspace (5) and attached to a compressor disk (3) with their radial outward end section, wherein the end section of the secondary air tubes (2) are attached to a separate carrier ring (6), which is connected to a compressor disk (3).

10

(Fig. 3)